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# Aviation fuel gauging sensor utilizing multiple diaphragm sensors incorporating polymer optical fiber Bragg gratings

C. A. F. Marques, A. Pospori, D. Sáez-Rodríguez, K. Nielsen, O. Bang, D. J. Webb, *Member, IEEE*

**Abstract** — A high performance fuel gauging sensor is described that uses five diaphragm-based pressure sensors, which are monitored using a linear array of polymer optical fiber Bragg gratings. The sensors were initially characterized using water, revealing a sensitivity of 98 pm/cm for four of the sensors and 86 pm/cm for the fifth. The discrepancy in the sensitivity of the fifth sensor has been explained as being a result of the annealing of the other four sensors. Initial testing in JET A-1 aviation fuel revealed the unsuitability of silicone rubber diaphragms for prolonged usage in fuel. A second set of sensors manufactured with a polyurethane based diaphragm showed no measurable deterioration over a 3 month period immersed in fuel. These sensors exhibited a sensitivity of 39 pm/cm, which is less than the silicone rubber devices due to the stiffer nature of the polyurethane material used.

**Index Terms**— Fiber Bragg gratings, Polymer optical fiber sensors, Pressure sensors, Fuel level monitoring, Aircraft applications.

## I. INTRODUCTION

FUEL level monitoring has always been a technical challenge. Generally, in aircraft fuel systems [1] the most frequently used level sensors are the capacitive and ultrasonic

devices, however they suffer from intrinsic safety concerns in explosive environments combined with issues relating to reliability and maintainability. In recent years, many optical fiber liquid level sensors have been reported to be safe and reliable and present many advantages for aircraft fuel measurement [2]. Above all, water mixed in the fuel will have little influence on optical fiber based liquid level sensors. Different optical fiber liquid level sensors have been developed, such as the pressure type, float type, optical radar type, total internal reflection type and side-leaking type [1,3-5]. Amongst these, many types of liquid level sensors based on fiber gratings have been demonstrated [6–11], including fiber Bragg gratings (FBGs), long period gratings, and tilted FBGs. However, these sensors have not been commercialized because they exhibit some drawbacks, such as low sensitivity, limited pressure range, long-term instability, limited resolution, high cost, weakness, and are complicated to manufacture. In addition, any sensors that involve direct interaction of the optical field with the fuel (either by launching light into the fuel tank or via the evanescent field of a fiber-guided mode) must be able to cope with the potential build up of contamination – often bacterial – on the optical surface.

We recently proposed an alternative optical approach utilizing multiple diaphragm-based pressure sensors, where the diaphragm deflection was monitored by embedded polymer optical fiber Bragg grating strain sensors [12]. The sensors are placed at different heights in the liquid tank (see Fig. 1) and the level determined by linear regression using the readings from the submerged sensors. This approach has several advantages:

- i. Fault tolerance: malfunctioning sensors can be identified and their outputs ignored;
- ii. Operation independent of fuel density: changing the density alters the slope of the fitted line, but not its intercept;
- iii. Operation insensitive to g-force: this again changes the slope of the line but not its intercept;
- iv. Temperature insensitivity: temperature induced shifts in both the nominal Bragg wavelengths of the sensors and the sensitivity of the sensors are compensated for.

It should be noted that the aircraft fuel gauging problem is particularly demanding since not only can the effective g-force vary due to acceleration, but the attitude of the plane to the

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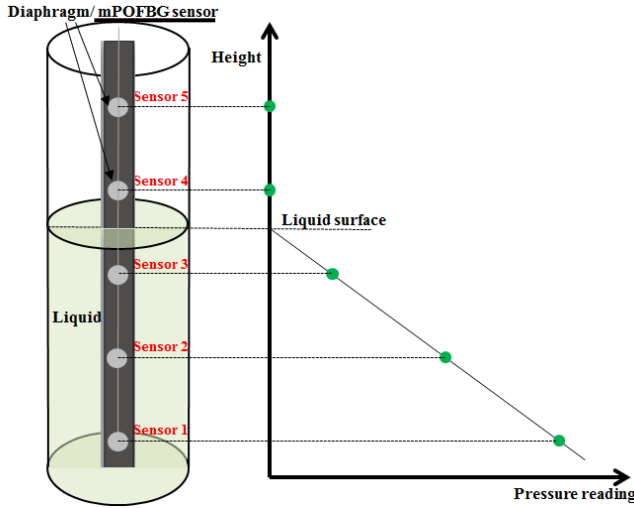


Fig. 1. Liquid level sensing approach using multiple pressure sensors. Left: sensor locations. Right: Level determination by linear regression to readings from submerged sensors.

effective gravitational force also change: in other words the plane of the liquid surface can have different orientations with respect to the airframe structure. This is a problem common to almost all gauging systems and is solved by having multiple level gauges coupled with appropriate signal processing.

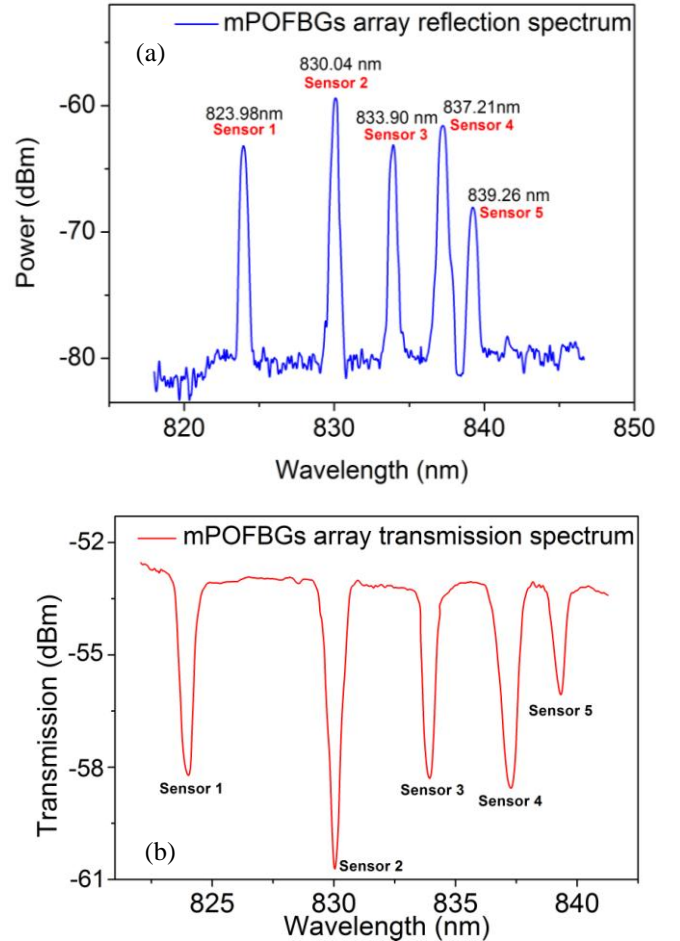
In this paper we build on the basic, proof-of-principle work reported in [12] where water was used as the liquid to be gauged and [13] where the response of the sensor was modeled. We report on experiments that reveal important engineering problems with the approach, for which we provide some solutions. Firstly we discover and characterize the role that annealing has on the sensor stress sensitivity. Secondly, we perform the first tests of our system using Jet A-1 aviation fuel; to enable long-term use in this environment we move to a polyurethane diaphragm, resistant to the fuel.

## II. PROPOSED SENSOR: FABRICATION AND SENSING PRINCIPLE

Five Bragg gratings were inscribed in a microstructured polymer optical fiber (mPOF) fabricated from poly(methyl methacrylate) (PMMA) and doped with benzyl dimethyl ketal (BDK) photoinitiator – for details of the fabrication see [14]. The mPOF has a core diameter of 6  $\mu\text{m}$  and an outer diameter of 125  $\mu\text{m}$  and a loss of about 7 dB/m at 850 nm. Using a single 75 cm long fiber, the five multiplexed mPOFBGs are inscribed spatially separated by 15 cm using a CW He–Cd laser with an output power of 30 mW at 325 nm. The inscription process was monitored using a continuous wave, super-luminescent diode from Superlum centered at 835 nm (with a power output of 1.25 mW over a spectrum width of 50 nm) with FC/APC terminated pigtail and an optical spectrum analyzer (OSA) connected to an 850 nm single-mode silica fiber coupler. To obtain five gratings with different wavelengths, two different phase masks were used with pitches of 557.5 and 580 nm and thermal annealing of the inscribed fiber was used to change the Bragg wavelengths [15]. The reflection and transmission spectra of the sensors are

shown in Figs. 2 (a) and (b), respectively. The nominal sensor wavelengths are 823.98 nm (sensor 1), 830.04 nm (sensor 2), 833.90 nm (sensor 3), 837.21 nm (sensor 4) and 839.26 nm (sensor 5). Following grating inscription, the mPOF containing the FBGs was UV-glued (Loctite 3936) to one 8° angled silica fiber pigtail. This multiplexed array of POFBGs is the largest reported to-date, indicative of the maturing nature of the technology.

For the diaphragm manufacture, a silicone rubber solution was prepared by mixing homogeneously two liquids [16] (SILASTIC® T-4 Base and Catalyst from DowCorning Corporation) in a ratio of 100:10 by volume. The prepared silicone rubber solution was poured in a 50 mm diameter plastic container with a height of 1.1 mm (see Fig. 2 (c)), in which was also placed the POF containing one of the FBGs. The diaphragm fabrication is a key step in the sensor fabrication and considerable care was taken to ensure the mPOFBG was at the center of each diaphragm (see Figs. 2 (c) and (d)) to obtain the same sensitivity in each sensor. With regard to uniformity, the diaphragms obtained had thicknesses around  $1.08 \pm 0.01$  mm, as measured using digital calipers. Calculations were made to obtain the same volume of silicone rubber solution in each diaphragm. The mold was kept undisturbed for 24 hours at room temperature to allow the silicone rubber to set. More details of the fabrication method can be found in [12].



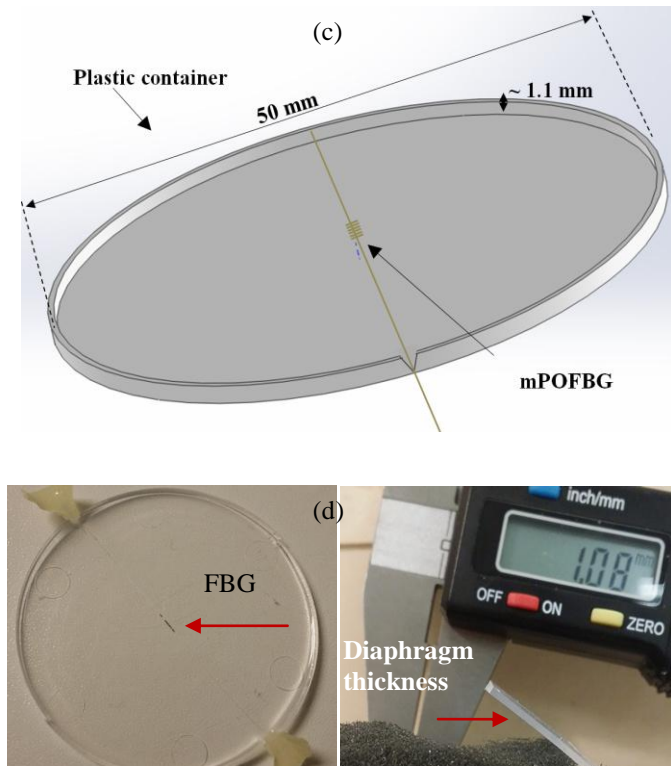


Fig. 2. (a) Reflection and (b) transmission spectra of mPOFBG array with five multiplexed gratings. (c) Plastic container used for the diaphragm fabrication and (d) an image of the silicone rubber diaphragm.

The design of the prototype multiple sensor system consists of a square acrylic tube (800 mm length, with 3.2 mm wall thickness and 38.1 mm outside dimension), with windows drilled at equidistant positions along it as shown in Fig. 3. It contains five sensors positioned over 15 mm diameter holes spatially separated by 150 mm. A judgment was made that 5 sensors were sufficient to demonstrate the proof of principle of our level sensing approach and allow evaluation. In an engineering application the number of sensors would be a design parameter to be optimized, since more sensors would improve system performance but increase complexity and therefore cost. The sensors were then placed and sealed at positions aligned with the window positions such that the FBG center was aligned with the window center. A thin layer of silicone sealant was used to seal effectively the sensing area and a slight strain was applied to the diaphragm when it was sealed to avoid hysteresis effects. Furthermore, a retaining ring was used in each sensor, with the diaphragm sandwiched between the tube and retaining ring. Eight screws were used to hold the tube and retaining ring together, producing a strong seal (see Fig. 3 (a)). With the tube sealed at the bottom, and open at the top, the atmospheric pressure inside the tube remains essentially constant. The system relies on increasing hydrostatic pressure from liquid outside the sensing tube deforming the diaphragms causing the fiber to become elongated, which results in a positive shift in the Bragg wavelength; otherwise, the internal pressure matches the external pressure and the diaphragm is not deformed.

### III. EXPERIMENTAL PROCEDURE/DETAILS

The experimental setup for evaluation of the liquid level sensitivity of the sensor is depicted in Fig. 3. The sensor responses were monitored and the data were collected with the super-luminescent diode and OSA used to monitor the inscription process. When the liquid level in the container rises, we deform the submerged diaphragms and consequently induce a positive shift in the Bragg wavelength is achieved. Some initial tests of the response time of the sensors were carried out before collecting the data to determine liquid level sensitivity.

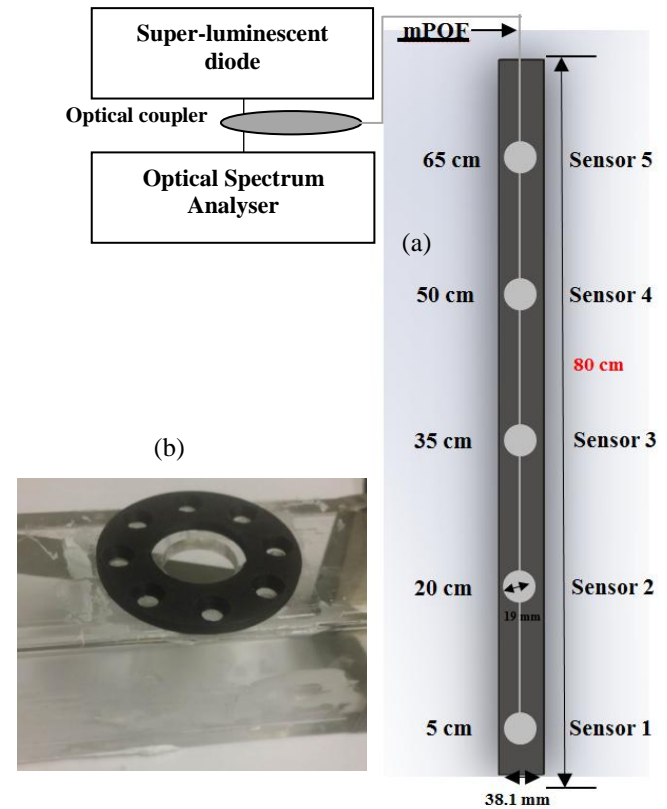


Fig.3. (a) Experimental apparatus and diagram of the acrylic tube sensor arrangement using mPOFBG array sensors. (b) Fixation of sensors to the square acrylic tube with retaining rings.

### IV. RESULTS AND DISCUSSION

The system was first characterized using water. The sensor performance was tested within a liquid level range of 0 to 75 cm and with a liquid level increment step of 2.5 cm. Two experiments using different diaphragm/mPOFBG sensors were carried out, and three cyclic tests were performed to investigate both increasing and decreasing levels of the liquid to check for any hysteresis in the behavior of the sensors. It should be noted that all experiments showed a very good agreement between them, proving good repeatability between sensor responses. The results of the three cycles of each sensor are summarized in Table 1. Figs. 4 (a) and (b) show the first cycle of the first experiment for sensor 1 and sensor 3, respectively. The wavelength shift was extracted and the sensitivity of each sensor was calculated, showing a sensitivity



of  $98.6 \pm 0.3$  pm/cm (sensor 1),  $98.1 \pm 0.2$  pm/cm (sensor 2),  $98.4 \pm 0.6$  pm/cm (sensor 3),  $97.6 \pm 0.8$  pm/cm (sensor 4), and  $86.1 \pm 2.6$  pm/cm (sensor 5). The insets in Figs. 4 (a) and (b) depict the residuals of each cycle (sensor 1 and sensor 3), showing a departure from linearity usually less than 0.1 nm, predominantly arising as a result of a small amount of non-linearity in the response. In terms of sensitivity variation between each sensor, one can see that there is a significant discrepancy coming from sensor 5. This was the only sensor where the fiber was not annealed following inscription and previous work had suggested that annealing led to a slight increase in the strain sensitivity of a POFBG recorded in a step index fiber [17]. Consequently, we carried out an investigation of the effects of annealing on the mPOF used for our sensors. In addition to studying the strain sensitivity, we investigated for the first time the annealing dependence of the force and stress sensitivity of the grating. The force sensitivity is important in this application because, due to its elastic modulus being higher than the diaphragm material, the POF will tend to restrict the pressure-induced expansion of the diaphragm. It is probably better then to think of the POFBG as being subjected to a pressure dependent force rather than having a pressure dependent strain imposed on it.

TABLE I  
THREE CYCLES RESULTS OF ALL SENSORS: SENSITIVITY ANALYSIS.

|                 | Increasing (pm/cm) | Decreasing (pm/cm) |
|-----------------|--------------------|--------------------|
| <b>Sensor 1</b> |                    |                    |
| Cycle 1         | $98.4 \pm 0.3$     | $99.1 \pm 0.4$     |
| Cycle 2         | $98.7 \pm 0.2$     | $98.9 \pm 0.5$     |
| Cycle 3         | $98.5 \pm 0.4$     | $98.6 \pm 0.2$     |
| <b>Sensor 2</b> |                    |                    |
| Cycle 1         | $98.1 \pm 0.2$     | $98.3 \pm 0.4$     |
| Cycle 2         | $98.4 \pm 0.2$     | $98.5 \pm 0.3$     |
| Cycle 3         | $98.7 \pm 0.4$     | $98.2 \pm 0.6$     |
| <b>Sensor 3</b> |                    |                    |
| Cycle 1         | $98.2 \pm 0.6$     | $98.7 \pm 0.5$     |
| Cycle 2         | $98.7 \pm 0.3$     | $99.2 \pm 0.6$     |
| Cycle 3         | $98.9 \pm 0.4$     | $98.8 \pm 0.3$     |
| <b>Sensor 4</b> |                    |                    |
| Cycle 1         | $97.4 \pm 0.8$     | $97.8 \pm 0.7$     |
| Cycle 2         | $98.2 \pm 0.9$     | $97.9 \pm 1.0$     |
| Cycle 3         | $97.9 \pm 0.6$     | $98.6 \pm 0.8$     |
| <b>Sensor 5</b> |                    |                    |
| Cycle 1         | $86.1 \pm 2.5$     | $86.2 \pm 2.7$     |
| Cycle 2         | $86.4 \pm 2.3$     | $86.0 \pm 2.2$     |
| Cycle 3         | $86.5 \pm 2.1$     | $86.1 \pm 2.4$     |

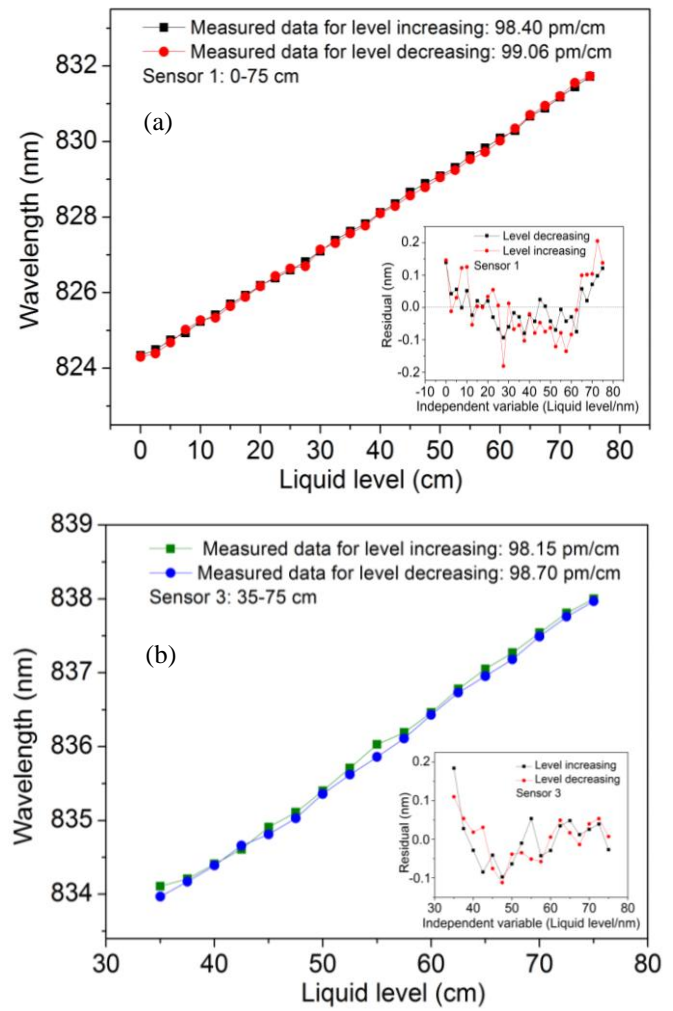


Fig. 4. Cyclic response using water for: (a) sensor 1 and (b) sensor 3. Insets: Residual plots for sensor 1 and sensor 3, respectively.

A grating with a nominal Bragg wavelength of 831.2 nm was fabricated in the doped mPOF used in the construction of the pressure sensors. The grating was attached using magnetic clamps between a fixed support and a translation stage enabling a controlled strain to be applied over a gauge length of 19.75 cm. The fiber was stretched in steps of 100 microns up to a total strain of 0.5% over a period of approximately 7 min, while the Bragg wavelength was monitored using the broad band source and OSA used previously. The experiment was repeated three times to ensure the results were repeatable. Following the strain experiments, the sensitivity of the grating to force was assessed by hanging the fiber vertically from a support and attaching a mass of  $2.990 \pm 0.005$  g. The wavelength change resulting from the application of the weight was determined (again, this was repeated 3 times to ensure reliable data). Finally the fiber diameter was measured using a microscope to enable the applied stress to be calculated.

Having completed this set of experiments, the fiber was annealed by placing the grating in water at  $64^\circ\text{C}$  for 2 minutes. This resulted in a permanent reduction of the Bragg

wavelength of 1.3 nm. Following this annealing process, the fiber was again characterized for strain, force and stress sensitivity as previously described. The results are summarized in Table 2, and are typical of those obtained for several other POFBGs fabricated in this fiber.

The amount of annealing carried out here is quite small compared the wavelength shifts induced in the POFBGs used for the pressure sensors. Nevertheless it may be seen from Table 1 that the annealing has a significant effect on the POFBG behavior, increasing the strain, force and stress sensitivity. We believe this process is responsible for the discrepancy observed in the pressure characteristics of sensor 5 in the level sensing experiment. To emphasize that, a very recent paper [18] shows that PMMA mPOFBGs annealed at high humidity have a superior response with a very low strain hysteresis, an improved strain sensitivity, and an increased stable operation temperature range.

TABLE II  
EFFECTS OF ANNEALING ON POFBG SENSITIVITY TO STRAIN AND STRESS

|   | Before annealing  | After annealing   |
|---|-------------------|-------------------|
| Strain sensitivity (pm/ $\mu\epsilon$ ) | $0.68 \pm 0.01$   | $0.73 \pm 0.02$   |
| Force sensitivity (pm/ $\mu\text{N}$ )  | $0.109 \pm 0.001$ | $0.137 \pm 0.001$ |
| Fiber diameter ( $\mu\text{m}$ )        | $128 \pm 2$       | $142 \pm 2$       |
| Stress sensitivity (pm/kPa)             | $0.14 \pm 0.02$   | $0.22 \pm 0.02$   |

Similar pressure array sensors were then fabricated to undertake the first tests using JET A-1 aviation fuel. This fuel presents a density around 0.810 kg/L at 15 °C (less than water density, which is around 0.999 kg/L at 15°C [19]). Two sensors were used, labeled sensors 1 and 3, as shown in Fig. 5 (a). The fuel depth was varied between 35 cm and 75 cm in steps of 5 cm, giving a 40 cm measurement region as shown in Fig. 5 (a). Two cyclic tests were performed to investigate both increasing and decreasing levels of the liquid. Figs. 5 (b) and (c) show the first cycle of sensors 1 and 3, respectively. The wavelength shift was extracted and the sensitivity of each sensor was calculated, showing a sensitivity of  $96.7 \pm 0.3$  pm/cm (sensor 1) and  $97.8 \pm 0.6$  pm/cm (sensor 3). These are quite similar sensitivities to those obtained with water, whereas we might expect the results with fuel to be about 20% lower due to the reduced density. The explanation for this apparent discrepancy probably relates to the use of a different sealant for fixing the diaphragm to the fuel container.

The determination of liquid level was also made by linear regression to the wavelength shift from the sub-surface sensors at different depths. The liquid surface level is set at  $48.0 \pm 0.1$  cm. The result is depicted in Fig. 5 (c), showing the equation of the linear fit. From the equation presented in Fig. 5 (c), an intercept value of  $46.69 \pm 1.06$  cm was computed. This proves for the first time that our sensing system is not affected by changes in the density of the liquid being monitored.

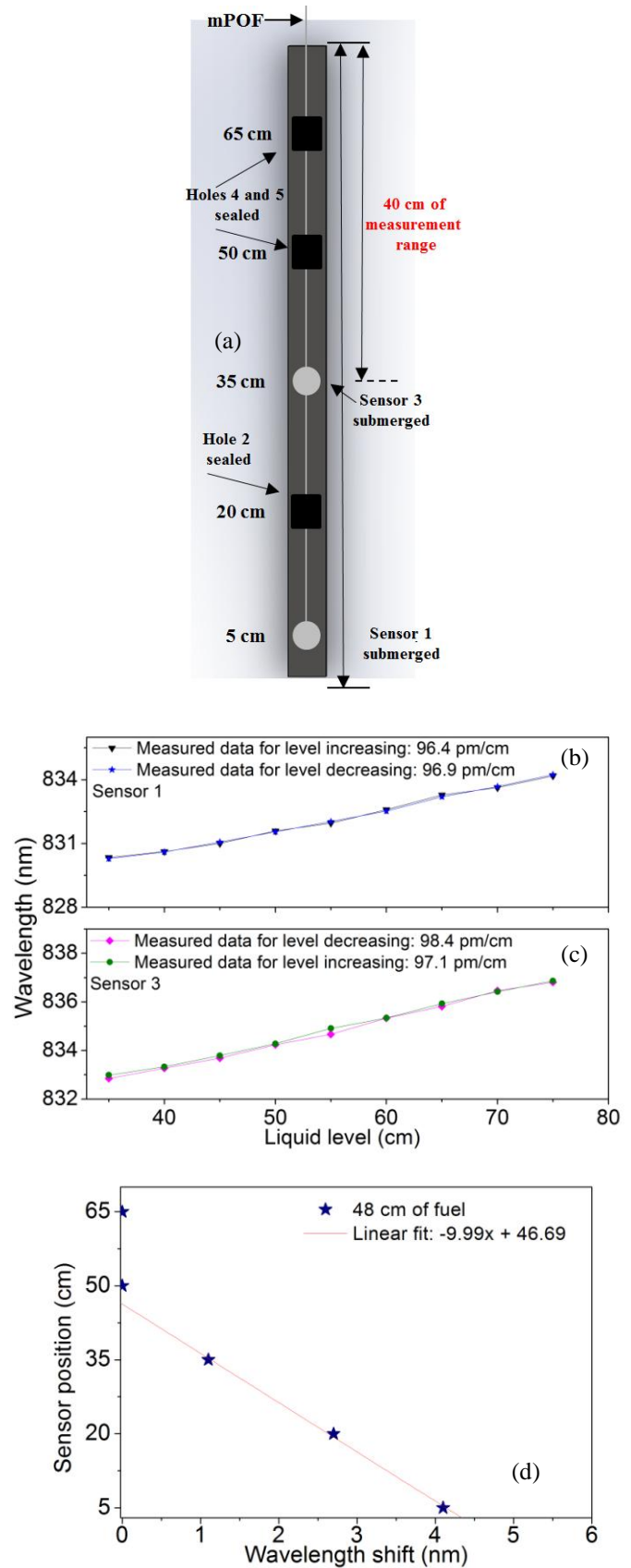


Fig. 5. (a) Diagram of the acrylic tube sensor arrangement showing the sensors submerged and the measurement range using fuel. Cyclic response using fuel for: (b) sensor 1 and (c) sensor 3. (d) Determination of liquid level using linear regression for a position of the fuel surface at 48 cm.

Despite these initially promising results with Jet A-1, the SILASTIC diaphragms were found to be unsuitable for long-term measurements. After a few hours exposed to fuel, the diaphragms were observed to swell to almost twice their original size, see Fig. 6 (a). Consequently a search was made for a more suitable material and a polyurethane resin from Liquid Lens [20] was chosen. This material is based on a mixing of two liquids - MF633 resin and DK780 catalyst - in a ratio of 100:100 by volume and curing at room temperature in 8 hours (23°C, 40% RH). This material shows less flexibility than the previous one, however it is resistant to the fuel. A diaphragm with the same dimensions as the previous manufactured diaphragms was placed in JET A-1 for more than 3 months with no measurable change in dimensions (see Fig. 6 (b)).

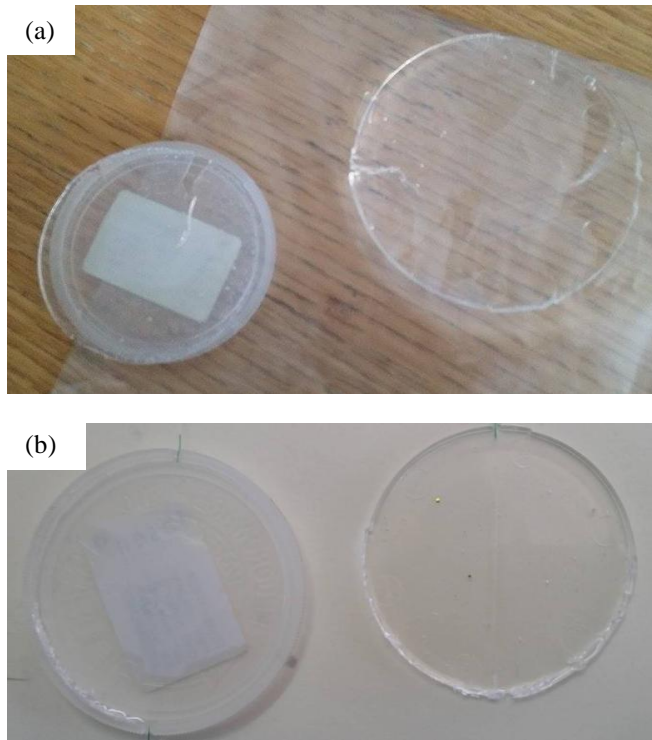


Fig. 6. (a) Diaphragms fabricated from SILASTIC. Left: as fabricated. Right: after exposure to fuel for several hours. (b) Diaphragms produced from polyurethane resin: left: as fabricated. Right: after exposure to fuel for more than 3 months with no measurable change in dimensions.

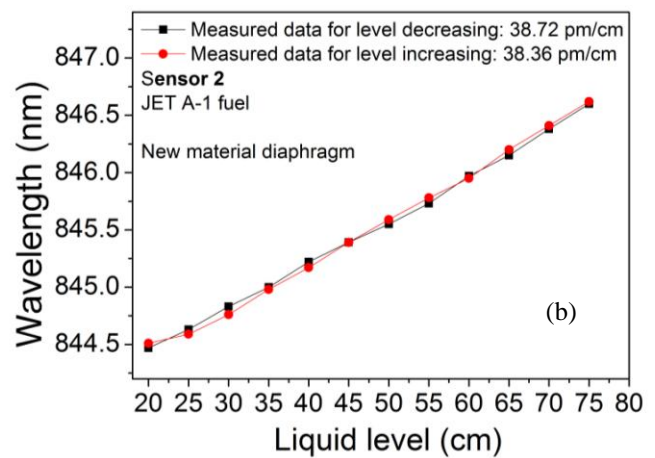
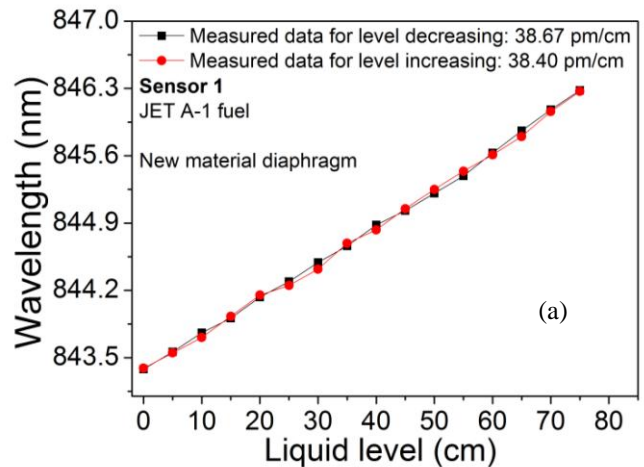
Sensors embedded in polyurethane were produced using the same procedure as earlier. The fuel level monitoring capability was tested within a liquid level range of 0 to 75 cm and with a liquid level increment step of 5 cm. We conducted 2 series of experiments using different sensors. For each experiment, three cyclic tests were performed to investigate both increasing and decreasing levels of the liquid to check for any hysteresis in the behavior of the sensors. All cyclic tests showed a very good agreement and Figs. 7 (a-c) show the first cycle of the first experiment for sensor 1, sensor 2 and sensor 3, with the position of each sensor indicated in Fig. 3. The results of the three cycles of each sensor are summarized in Table 3. The wavelength shift was extracted and the sensitivity of each sensor was calculated, showing an average sensitivity of  $38.5 \pm 0.5$  pm/cm. The achieved sensitivity is

TABLE III  
THREE CYCLES RESULTS OF ALL SENSORS: SENSITIVITY ANALYSIS.

|                 | Increasing (pm/cm) | Decreasing (pm/cm) |
|-----------------|--------------------|--------------------|
| <b>Sensor 1</b> |                    |                    |
| Cycle 1         | $38.4 \pm 0.5$     | $38.7 \pm 0.4$     |
| Cycle 2         | $38.7 \pm 0.2$     | $38.5 \pm 0.3$     |
| Cycle 3         | $38.5 \pm 0.4$     | $38.6 \pm 0.2$     |
| <b>Sensor 2</b> |                    |                    |
| Cycle 1         | $38.4 \pm 0.4$     | $38.7 \pm 0.6$     |
| Cycle 2         | $38.2 \pm 0.4$     | $37.4 \pm 0.3$     |
| Cycle 3         | $37.5 \pm 0.2$     | $38.6 \pm 0.4$     |
| <b>Sensor 3</b> |                    |                    |
| Cycle 1         | $38.8 \pm 0.2$     | $39.5 \pm 0.4$     |
| Cycle 2         | $38.4 \pm 0.5$     | $38.6 \pm 0.4$     |
| Cycle 3         | $38.5 \pm 0.3$     | $38.3 \pm 0.4$     |

less than with the SILASTIC silicone rubber due to the increased stiffness of the new material.

Once again, the determination of liquid level was made by linear regression to the wavelength shift from the sub-surface sensors at different depths. The liquid surface level is set at  $48.0 \pm 0.1$  cm. Fig. 7 (d) shows the equation of the linear fit. From the equation presented, an intercept value of  $47.5 \pm 1.0$  cm was computed.





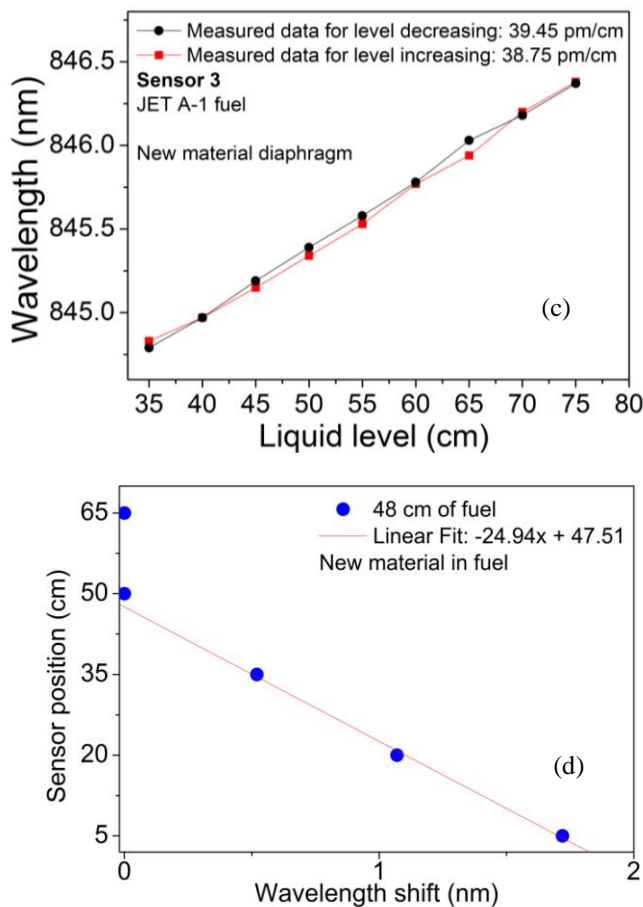


Fig. 7. Cyclic response using new diaphragm material (polyurethane resin from Liquid Lens [19]) in contact with JET A-1 for: (a) sensor 1, (b) sensor 2 and (c) sensor 3. (d) Determination of fuel level using linear regression for a position of the fuel surface at 48 cm.

## V. CONCLUSION

For the first time a high performance liquid level sensor based on a single fiber, multiplexed array of mPOFBG sensors embedded in silicone rubber diaphragms was fabricated and its performance studied in detail. The experimental results show that the proposed system had a high sensitivity to liquid level, and exhibited a highly linear and repeatable response. This mPOFBG array sensor, when compared with the best approach based on silica optical fiber published in the literature, exhibits a factor of 4 improvement in sensitivity (98 pm/cm). The level sensor has been used with JET A-1 aviation fuel and whilst initial performance was good, the tests revealed the diaphragm material to be unsuitable for prolonged use. An alternative polyurethane material has been successfully employed which shows good promise for long term use in fuel, albeit at the expense of a reduced sensitivity, due to the larger elastic modulus of the material.

Finally, initial investigations into the role of annealing on the sensing behavior of POFBGs has been undertaken, revealing that the strain, force and stress sensitivity of the devices are increased by just a modest amount of annealing.

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